

#### Exploring and Exploiting Rarefied Flows for Novel Microdevices

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# **Applications of Rarefied Gas Dynamics**



# Outline

- Exploring low-speed microflows:
  - Deterministic methods for rarefied gas flows:
    - A-priori accuracy estimate: discrete H-theorem
    - FVM + Immersed boundary / cut-cell methods for moving boundaries
    - Discontinuous Galerkin methods
  - Example: understanding rarefied gas damping effects in RF MEMS switches by coupled fluid/structural/electrostatic analysis
- Exploiting microscale effects for microdevices:
   Knudsen compressor for on-chip vacuum
   Knudsen force actuation/sensing
   Field-emission driven microdischarges

### **Gas-Based Microdevices**



MIT micro-turbine

USC micro-burner

NASA/UVM micro-thruster

#### **Microscale Challenges:**

Gas-phase extinction limit, min Re~40. Catalyst needed
Amplified heat transfer losses
Increased viscous losses: Isp drops for Re<200.</li>

Rarefied flow analysis provides methods for design to overcome these challenges

# Microdevices Exploiting Rarefied Flow

# Vargo, Muntz , Shiflett, Tang, "Knudsen Compressor as a Micro/Meso-scale Vacuum Pump", *JVST A*, 1999:

a cascade of multiple, individually heated compressor stages that exploit the pumping effect of rarefied thermal transpiration.



<u>1-stage</u>: 1.5 W, 2 cmx2cm 760 → 750 torr  $\frac{1-\text{stage}}{80 \text{ mW}}$ : 80 mW, 2 x 2 mm<sup>2</sup> 760 → 350 torr

# Numerical Modeling Approach: FVM

Unsteady Boltzmann model kinetic equation for velocity distribution function:

$$\frac{\partial f}{\partial t} + \vec{u} \cdot \nabla_{\vec{x}} f = \frac{1}{\tau} (f_0 - f)$$

•ESBGK collision operator

- $\succ$  H-theorem proved by Andries et al, 2000.
- Conservative discretization by perturbed Gaussian by Mieussens, JPC, 2000.

#### FMV Solver:

•Discrete ordinates in velocity space with Gauss-Hermite

- •FVM in physical space with 3<sup>rd</sup>-order WENO fluxes.
- •2<sup>nd</sup> and 3<sup>rd</sup> -order time integration with Runge-Kutta TVD schemes
- domain decomposition in physical space

### Entropy Properties

Boltzmann's Entropy: S: Boltzmann's H-Theorem:  $\dot{S}$ :

$$S = -k \ln(\Omega)$$
$$\dot{S} \ge 0$$

$$S = k \int_{-\infty}^{\infty} f(\vec{c}) \left[ 1 - \ln\left(\frac{h^3 f(\vec{c})}{m^3}\right) \right] d\vec{c}$$

High entropy generation rate as indicator of nonequilibrium (Naterer & Camberos, JTHT, 2003)

Entropy generation rate as moment of BE:

$$\begin{split} \dot{S}_{lhs} &= \frac{\partial S}{\partial t} + \nabla \cdot \left( k \int_{-\infty}^{\infty} \vec{c} f(\vec{c}) \left[ 1 - \ln \left( \frac{h^3 f(\vec{c})}{m^3} \right) \right] d\vec{c} \right) \\ \dot{S}_{coll} &= -\nu \int_{-\infty}^{\infty} \left( f(\vec{c}) - f_0(\vec{c}) \right) \ln \left( \frac{h^3 f(\vec{c})}{m^3} \right) d\vec{c} \end{split}$$

Entropy generation rate based on macro-parameters:

$$\dot{S} = \frac{\Phi}{T} + \frac{\kappa}{T^2} \nabla T^2 \qquad \Phi = \mu \left( \frac{2}{3} \left[ \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right]^2 + \frac{2}{3} \left( \frac{\partial u}{\partial x} - \frac{\partial v}{\partial y} \right)^2 + \left( \frac{\partial u}{\partial y} + \frac{\partial v}{\partial x} \right)^2 \right)$$



### **Accuracy Analysis: Discrete H-Theorem**

Governing equation:

$$\frac{\partial f}{\partial t} + \vec{c} \cdot \frac{\partial f}{\partial \vec{x}} = \frac{f_0 - f}{\tau}$$

Discrete equation:

$$L^{t}(f^{i,j}) + L^{x}(f^{i,j}) = \frac{f_{0}^{i,j} - f^{i,j}}{\tau^{i,j}}$$

Boltzmann's H-Theorem:

 $\dot{S} \ge 0$ 

Discrete analog of H-Theorem:

$$\dot{S}_{\text{lhs}}^{i,j} = L^{t}(S^{i,j}) + L^{x}\left(\sum_{j} c_{j} f^{i,j} \left[1 - \ln\left(\frac{h^{3} f^{i,j}}{m^{3}}\right)\right] \delta c_{j}\right)$$
$$\dot{S}_{\text{coll}}^{i,j} = -\nu \sum_{j} \left(f^{i,j} - f_{0}^{i,j}\right) \ln\left(\frac{h^{3} f^{i,j}}{m^{3}}\right) \delta c_{j}$$
$$S_{\text{lhs}}^{i,j} = S_{\text{coll}}^{i,j} \ge 0$$

Chigullapalli, Venkattraman, Ivanov, Alexeenko, J. Comp. Phys, 2010

Approach to equilibrium: 0D3V



## **1D3V Riemann Problem**

1<sup>st</sup> order





- Accuracy of numerical solution manifested in kinetic entropy generation rate.
- The entropy generation rate displays three peaks corresponding to the shock, contact discontinuity and the rarefaction wave.

Chigullapalli, Venkattraman, Ivanov, Alexeenko, J. Comp. Phys. 2010

### 1D3V M=1.4 Normal Shock



The entropy generation rate as indicator of compatibility of BC.

Chigullapalli, Venkattraman, Ivanov, Alexeenko, J. Comp. Phys, 2010

## BM for Microflows: Motivation

#### High Frequency Micro Electro Mechanical Devices (MEMS)



Raman Group -Fluid structure interaction and nonlinear dynamics in RF-MEMS devices (part of the PRISM center for the prediction of reliability, integrability, and survivability of microsystems) Purdue University



\*Czarnecki et.al, 19th IEEE Conference on MEMS, 2006

## **IBM Formulations: Interpolation**



Step 1: Find foutgoing using 2<sup>nd</sup> order Least Squares Interpolation. Find  $\beta = (M^T M)^{-1} M^T \Phi$ Step 2: Conservation of mass flux at solid face  $\sum_{\vec{c},\vec{n}>0} \vec{c}.\vec{n}f_{s,out} + n_{wall} \sum_{\vec{c},\vec{n}<0} \vec{c}.\vec{n}\exp(-\frac{(\vec{c}-\vec{v})^2}{T}) = 0$ Step 3: Interpolate fincoming at IB face

Step 4: Solve Aof+R=0 for all directions

### **IBM Formulations: Relaxation**

Step 1: Relax  $f_{IB}$  for each direction using  $f_{s,out} = f_{\gamma,FCells} + (f_{IB,FCells} - f_{\gamma,FCells}) \exp(\frac{-t}{\tau_{IB}})$ 

$$t = \frac{\Delta x_{IB}}{\vec{c}.\vec{n}} [\text{Ref. 4,5}]$$

Step 2: Conservation of mass flux at solid face centroid to find f<sub>s,in</sub>

$$\sum_{\vec{c}.\vec{n}>0} \vec{c}.\vec{n}f_{s,out} + n_{wall} \sum_{\vec{c}.\vec{n}<0} \vec{c}.\vec{n}\exp(-\frac{(\vec{c}-\vec{v})^2}{T}) = 0$$

Step 3: Relax f<sub>s,in</sub> for each directions using

$$f_{IB,in} = f_{\gamma,IB} + (f_{s,out} - f_{\gamma,IB}) \exp(\frac{-t}{\tau_{IB}})$$



C.K. Chu Kinetic theoretic description of the formation of shock ware, Phys. Of Fluids, 8(1):12, 1965 Y. Ruan and A. Jameson, Gas-Kinetic BGK schemes for three-dimensional compressible flows, AIAA 2002-0550

### **IBM Fluxes: Interrelaxation**

Step 1: Interpolate macroparameters and f at IB face using only interior values. Interpolation Relaxation

Step 2: Relax f<sub>IB</sub> for each direction using

$$f_{s,out} = f_{\gamma,IB} + \left(f_{IB,out} - f_{\gamma,IB}\right) \exp\left(\frac{-t}{\tau_{IB}}\right) \quad t = \frac{\Delta x_{IB}}{\vec{c}.\vec{n}}$$

Step 3: Conservation of mass flux at solid face centroid to find fsin

$$\sum_{\vec{x},\vec{n}>0} \vec{c}.\vec{n}f_{s,out} + n_{wall} \sum_{\vec{c},\vec{n}<0} \vec{c}.\vec{n}\exp(-\frac{(\vec{c}-\vec{v})^2}{T}) = 0$$

Relaxation Step 4: Interpolate macroparameters at IB face using interior and solid

i-1

values

Step 5: Relax  $f_{s,in}$  for each j  $f_{IB,in} = f_{\gamma,IB} + (f_{s,out} - f_{\gamma,IB}) \exp(\frac{-\iota}{\tau_{IB}})$ 

Error is less than 1% when  $Kn_{\Delta x_{IB}} = \frac{\lambda}{\Delta x_{IB}} \ge 1.9$ 

#### **IBM-ESBGK: 2D** Verification



Solid - 10,800 cells; Dashed - 20,100 cells

### JBM-ESBGK: Microbeam under 1E6 g

#### Temperature

**Pressure** 





**∆t**= 6.25 ns

Max Velocity= 5.684 m/s

Width: 5  $\mu$ m, Width/Gap<sub>0</sub> = 2.5 Background mesh: 10,920 cells  $\Delta x$ =0.1  $\mu$ m,  $\Delta y$ =0.2 to 1  $\mu$ m

#### IBM-ESBGK: Microbeam under 1E6 g



Width: 5  $\mu$ m, Width/Gap<sub>0</sub> = 2.5 Background mesh: 10,920 cells  $\Delta x$ =0.1  $\mu$ m,  $\Delta y$ =0.2 to 1  $\mu$ m

# **3D3V Gas Damping in MEMS switch**

#### Pressure and Streamlines for PRISM Gen5 Device during Pull-In







# **PRISMCG Online Tool**



# Impulsive Effects in Switch Dynamics



Flexural wave reflection from boundaries lead to stresses which are higher by > 2.5x compared to static.

# Impulsive Dynamics and Switch Lifetime



- The maximum bending stress in the beam for a given voltage increases with decreasing pressure.
- The maximum stress is converted to cycles to failure using a S-N relation\* for typical LIGA Ni.

# High-Order Methods for Boltzmann-ESBGK

#### High-order deterministic methods

- Discretizations of Coordinate Space
  - Finite Difference Method & Finite Volume with high-order fluxes (i.e. WENO) Review by Luc Mieussens, RGD 2014.
  - High-order Finite Element / Discontinuous Galerkin methods Lesaint & Raviart, 1974
  - Spectral methods: work better on globally smooth functions

#### Advantages of Runge-Kutta discontinuous Galerkin method

- Very suitable for the solution of time-dependent hyperbolic and advection dominated advection-diffusion equations
- naturally obtain fluxes at the boundaries with the same high-order accuracy as in the interior of the domain
- Efficient parallel implementation due to compactness of the scheme

#### Additional references:

B. Cockburn and C. W. Shu. TVB Runge-Kutta local projection discontinuous Galerkin finite element method for scalar conservation laws IV: The multidimensional case. *Math. Comp.*, 54: 545-581,1990.

A. Alexeenko, C. Galitzine, A. M. Alekseenko, AIAA Paper 2008-4256

M. Alekseenko. Numerical properties of high order discrete velocity solutions of the BGK kinetic equation. *Applied Numerical Methermatics*, 61(4), 2011. pp 410-427

W. Su, A. Alexeenko, C. Cai, A Runge-Kutta discontinuous Galerkin solver for 2D Boltzmann model equations: Verification and analysis of computational performance, RGD 2012

W. Su et al, A Stable Runge-Kutta Discontinuous Galerkin Solver for Hypersonic Rarefied Gaseous Flows, RGD 2014.



In each triangle, solutions f<sup>j</sup><sub>p</sub> are sought in the finite element space of discontinuous functions

$$f_{p}^{j}(t,x,y) = \sum_{l=1}^{k} F_{p,l}^{j,l}(t)\varphi_{l}^{l}(x,y)$$

- Basis Functions: supported in each triangle and dependent on the geometry
  - 2nd order:  $\varphi_i^l(x, y) = a_0^l + a_1^l x + a_2^l y, \ l = 1, 2, 3$
  - **3**<sup>rd</sup> other:  $\varphi_i^l(x, y) = a_0^l + a_1^l x + a_2^l y + a_3^l x^2 + a_4^l y^2 + a_5^l xy, \ l = 1, 2, ..., 6$
- Degree of freedom

Tim

• Determined by the weak formulation of the governing system

$$\sum_{l=1}^{k} M_{ml} \frac{d}{dt} F_{p,i}^{j,l}(t) + \sum_{e \in \partial K_{i}} \int_{e} h_{e,K_{i}}(x,y,t) \varphi_{i}^{m}(x,y) d\Gamma - c_{x}^{j} \sum_{l=1}^{k} F_{p,i}^{j,l} Q_{ml}^{x} - c_{y}^{j} \sum_{l=1}^{k} F_{p,i}^{j,l} Q_{m}^{y}$$
  
=  $\int_{K_{i}} v \left( f_{E,p}^{j} - \sum_{l=1}^{k} F_{p,i}^{j,l} \varphi_{i}^{l}(x,y) \right) \varphi_{i}^{m}(x,y) dx dy \quad m = 1, ..., k$ 

Implicit Runge Kutta method up to 4<sup>th</sup> order

# **RKDG for Boltzmann-ESBGK**

Numerical Flux and Boundary Conditions:

The values of f<sup>j</sup><sub>p</sub> are discontinuous at the edges. Two-point Lipschitz numerical fluxes are used to replace the real fluxes

$$h_{e,K_i}(x,y,t) = \begin{cases} \boldsymbol{c}^{j} \cdot \boldsymbol{n}_{e,K_i} f_p^{j}(int(K_i),t), \ \boldsymbol{c}^{j} \cdot \boldsymbol{n}_{e,K_i} \ge 0\\ \boldsymbol{c}^{j} \cdot \boldsymbol{n}_{e,K_i} f_p^{j}(ext(K_i),t), \ \boldsymbol{c}^{j} \cdot \boldsymbol{n}_{e,K_i} < 0 \end{cases}$$

- The boundary values  $f_p^j$  (ext(K<sub>i</sub>),t) should be specified at the boundary edges
  - symmetry boundary
  - specular-diffuse moving wall with give accommodation coefficient
  - periodic boundaries,
  - far pressure inlet/outlet boundaries
  - supersonic inlet/outlet boundaries

<u>Conservative Discretizations of the Collision Term:</u>  $v(f_{E,p}^{j} - f_{p}^{j})$ 

- Specify the equilibrium distribution equation
  - BGK model  $f_{E,0}^{j} = \exp[a_1 a_2(c^j u) + a_3(c_x^j u) + a_4(c_y^j v)]$
- ES-BGK model  $f_j^j = \exp[a a(\vec{c}^j u) + a(c^j u) a(\vec{c}^j u)^2 + a(c^j v) + a(c^j u)^2 + a(c^j v)^2 + a(c^j$

$$a_s(x,y) = \sum_{i=1}^{n} A_s^{l} \varphi_i^{l}(x,y)$$

- the collision frequency and other macro properties can vary inside the spatial elements
- obtained from the weak formulation of mass, momentum and energy conservation for the collision relaxations

### **RKDG: 1D Heat Transfer**





## **RKDG: 1D Heat Transfer**

FVM-2

# RKDG-3

Case	CPU time, sec	$\mathrm{T}(\tfrac{x}{L}{=}0.5),\mathrm{K}$
N=8	8.8E-002	180.532
N=16	0.388	187.241
N=32	1.62	190.526
N=64	6.54	191.847
N=128	25.86	192.305
N=256	107.1	192.447
$N=512^{1}$	433.135	192.497

	Case	CPU time, sec	${\rm T}(\frac{x}{L}{=}0.5),{\rm K}$
	N=4	2.47	192.142, 192.487
<	N=8	10.24	192.476, 192.511
	N=16	45.24	192.5, 192.505
	N=32	193.98	192.501, 192.502
10	N=64	784.08	192.5, 192.5

# **RKDG: 2D Heat Conduction**

• 2D Conduction problem: *Kn*=0.0018 (comparison to analytical solution for heated lid cavity heat conduction)



# <u>RKDG:2D flow around thermal actuator</u>



## **RKDG: Automatic Mesh Refinement**

#### Coarse mesh in the lower gap Max Face Error = 0.002



#### Refined mesh in the lower gap Max Face Error = 0.0006



#### 11/03/14



## Exploiting Rarefied Flows for Novel Microdevices

# **Knudsen Thermal Force**



Crookes' Radiometer (Sir William Crookes, 1874)

$$U_{creep} \sim \lambda \cdot \frac{dT}{dx} = Kn \cdot \Delta T$$

#### Crookes Radiometer: Transverse



#### Knudsen Compressor: Longitudinal



## Kn force actuation

- Consequence of a thermal non-equilibrium between gas and solid
- Can be generated by resistive heating as well as optically Experimental data Passian et al, *PRL*, 2003 measurements using heated AFM probes

2.8

2.4

2.0

1.6

1.2

0.8

0.4

Signal [mV]

∱ F<sub>κn</sub>

Substrate



## **Passian et al data:** Simulation Results

ESBGK simulations for planar geometry with equivalent front-to-side area ratio

• Velocity Contours and Streamlines:



# **Experimental Validation**



- Argon and Nitrogen simulations agree with experiments within 10%.
- Deviation for Helium is about 80% at the maximum of Knudsen force.

#### Nabeth, Chigullapalli, Alexeenko, PRE, 2011

#### **Compact Model for Kn Force**

Closed-form model for non-dimensional Knudsen thermal force coefficient for uniformly heated beam:



## Force Enhancement and Reversal

• Uniformly heated beam (Passian et al, 2003)



• Thermoelectric heating for bi-directional actuation (this work):



## Modeling the Knudsen Force

Simulate Knudsen force on suspended heating element using ES-BGK model

- 2D-2V finite volume solver with second-order upwind fluxes
- 8<sup>th</sup>-order Gauss-Hermite quadrature in velocity space



### Simulation Results

#### Uniform Heating

Thermoelectric: Bottom-Up

#### Thermoelectric: Top-Down





# **Experimental Setup**





#### Purdue LEAP MicroNewton Thrust Stand



### Measurement Technique



- Measure displacement of reaction plate
- Sweep 0.5 3.0 mm gap size at fixed pressure
  - Hold for 60 seconds to ensure steady state
  - Apply linear drift correction

- Calibration performed at the beginning of each test day
- Sweep 0 100 V, 30 seconds high, 60 seconds low
  - Evaluate force from LVDT voltage



# Comparison of Modeling and Experiment



Maximum 36% error in C<sub>F,Kn</sub> between simulation and experiment
 Difference likely due to 3D effects in experimental measurements
 Errors likely stem from model inputs related to the heating element

New deterministic and stochastic kinetic approaches needed to explore and exploit significant new physics emerging at the micro/nanoscale:

- □ Low-speed
- □ Moving geometries
- Coupling to structural/thermal/EM solvers

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